The X-4 was designed to test a semi-tailless wing configuration at transonic speeds. Many engineers believed in the 1940s that such a design, without horizontal stabilizers, would avoid the interaction of shock waves between the wing and stabilizers. These were believed to be the source of the stability problems at transonic speeds up to Mach 0.9. Two aircraft had already been built using a semi-wingless design - the rocket-powered Me-163 Komet flown by Nazi Germany in World War II, and the British de Havilland DH.108 Swallow build after the war. The Army Air Forces signed a contract with the Northrop Aircraft Company on June 11, 1946, to build two X-4s. Northrop was selected because of its experience with flying wing designs, such as the N9M, XB-35 and YB-49 aircraft.

The resulting aircraft was very compact, only large enough to hold two J30 jet engines, a pilot,
instrumentation, and a 45-minute fuel supply. Nearly all maintenance work on the aircraft could be done without using a ladder or footstool. A person standing on the ground could easily look into the cockpit. For control without horizontal tail surfaces, the X-4 used combined elevator and aileron control surfaces (called elevons) for control in pitch and roll attitudes. The aircraft also had split flaps, which doubled as speed brakes.

The first X-4 (serial number 46-676) was delivered to Muroc Air Force Base, Calif., in November 1948. It underwent taxi tests, and made its first flight on December 15, 1948, with Northrop test pilot Charles Tucker at the controls. Winter rains flooded Rogers Dry Lake soon after, preventing additional X-4 flights until April 1949. The first X-4 proved mechanically unreliable, and made only 10 flights. Walt Williams, the head of the NACA Muroc Flight Test Unit (as Dryden was then known) called the aircraft a “lemon.”

The second X-4 (serial number 46-677) was delivered during the halt of flights, and soon proved far more reliable. It made a total of 20 contractor flights. Despite this, the contractor flight program dragged on until February 1950, before both aircraft were turned over to the Air Force and the NACA. The first X-4 never flew again, serving as a spare parts bin for the second aircraft.

The NACA instrumented the second X-4 to conduct a short series of flights with Air Force pilots. These included Chuck Yeager, Pete Everest, Al Boyd, Richard Johnson, Fred Ascani, Arthur Murray and Jack Ridley. The flights were made in August and September of 1950. The first flight by a NACA pilot was made by John Griffith on September 28, 1950.

The initial NACA X-4 flights, which continued from late 1950 through May of 1951, focused on the aircraft’s sensitivity in pitch. NACA pilots Griffith and Scott Crossfield noted that as the X-4’s speed approached Mach 0.88, it began a pitch oscillation of increasing severity, which was likened to driving on a washboard road. Increasing speeds also caused a tucking phenomena, in which the nose pitched down. More seriously, the aircraft also showed a tendency to “hunt” about all three axes. This combined yaw, pitch and roll, which grew more severe as the speed increased, was a precursor to the inertial coupling which would become a major challenge in the years to come.

To correct the poor stability, project engineers decided to increase the flap/speed brake trailing edge thickness. Balsa wood strips were added between the flap/speed brake halves, causing them to remain open at a 5 degree angle. The first test of the blunt trailing edge was flown on August 20, 1951, by NACA pilot Walter Jones. A second test was made by Crossfield in October. The results were positive, with Jones commenting that the X-4’s flight qualities had been greatly improved, and the aircraft did not have pitch control problems up to a speed of Mach 0.92.

The balsa strips were removed, and the X-4 then undertook a long series of flights to test landing
By opening the speed brakes, the lift-to-drag ratio of the aircraft could be reduced to less than 3 to 1. This was for data on future rocket powered aircraft. The tests continued through October 1951, until wing tank fuel leaks forced the aircraft to be grounded until March 1952, when the landing tests resumed. NACA pilots Joe Walker, Stanley Butchard, and George Cooper were also checked out in the aircraft.

The thickened flap/speed brake tests had been encouraging, so balsa wood strips were reinstalled on both the flap/speed brake and the elevons. The first flight was made by Jones on May 19, 1952, but one of the engines was damaged during the flight, and it was August before a replacement J30 could be found. When the flights resumed, they showed that the modifications had improved stability in both pitch and yaw, and delayed the nosedown trim changes from Mach 0.74 to Mach 0.91. Above Mach 0.91, however, the X-4 still oscillated.

In May of 1953, the balsa wood strips were again removed, and the X-4’s dynamic stability was studied in the original flap/speed brake and elevon configuration. These flights were made by Crossfield and John McKay. This was the final project for the X-4, which made its 81st and final NACA flight on September 29, 1953. Both aircraft survived the test program. The first X-4 was transferred to the Air Force Academy, Colorado Springs, Colo., before being returned to Edwards Air Force Base. The second X-4 went to the Air Force Museum, Wright-Patterson Air Force Base, Ohio.

The X-4’s primary importance involved proving a negative, in that a swept-wing semi-tailless design was not suitable for speeds near Mach 1. Aircraft designers were thus able to avoid this dead end. It was not until the development of computer fly-by-wire systems that such designs could be practical. Semi-tailless designs appeared on the X-36, Have Blue, F-117A, and Bird of Prey, although these aircraft all differed significantly in shape from the X-4.